SECTION I

LIMITATIONS

Airspeeds quoted in this section are Calibrated Airspeeds (CAS) unless otherwise noted.

Observance of the limitations listed is mandatory.

ENGINES

Take-off and Maximum Continuous Power	•	•	•	•	•	·	•	. 2900 rpm at 41.5 in. Hg
Airplane serials P-144 and after (and prior to with TIO-541-E1C4 Engines, one or both)				•				
Cruise Power above 27,000 ft					.2	2750	rpr	n and 31 in. Hg (Minimum)

Engine must be preheated before starting when ambient temperatures are below 10°F (-12°C).

Oil Cooler Baffles are to be removed when OAT exceeds 20°C.

PROPELLERS

Two Hartzell constant speed, full feathering, three-bladed propellers using; HC-F3YR-2 hubs with C7479-2R blades and C3273 spinner assemblies; Pitch setting at 30-inch station: Low, 14°; High, 81.7°. Diameter 74 inches, no cut-off permitted.

STARTERS

When restarting an engine in flight do not use the starter above 20,000 feet.

FUEL GRADE

100/130 (Green) Aviation Gasoline minimum grade 115/145 (Purple) Aviation Gasoline alternate grade.

FUEL CAPACITY

Baffled fuel cells installed in both wings (207 gallon capacity)	202 gallons usable after compliance with S.I. 0559-281
Unbaffled fuel cells in either or both wings (207 gallon capacity)	192 gallons usable after compliance with S.I. 0559-281

FUEL MANAGEMENT

Do not take-off if fuel quantity gages indicate in Yellow Arc or with less than 25 gallons of fuel in each main tank.

INSTRUMENT MARKINGS

Oil Temperature Minimum (Red Radial) Reference Serv Normal Operating Range (Green Arc) Maximum (Red Radial)												38 to 118°C
Oil Pressure Minimum Idle (Red Radial) Normal Operating Range (Green Arc) Maximum (Red Raidal)												60 to 90 psi
Fuel Flow Green Arc .	:	•	:	:	:	:	:	:	:	:	:	93 to 110 pph . 110 to 131 pph
Manifold Pressure Normal Operating Range (Green Arc) Maximun (Red Radial)												
Tachometer Normal Operating Range (Green Arc) Maximum (Red Radial)												2350 to 2900 rpm
Cylinder Head Temperature Normal Operating Range (Green Arc) Maximum (Red Radial)												
Turbine Inlet Temperature (Red Radial)						•			•			900°C
Instrument Air Caution Range (Yellow Arc) Normal Operating Range (Green Arc) Caution Range (Yellow Arc) Red Button Source Failure Indicators												
Normal Operating Range (Green Arc) . (Red Radial)	•	:								:		7 to 20 psi
Cabin Differential Normal Operating Range (Green Arc) Maximum (Red Arc)												. 0 to 4.7 psi
Propeller Anti-Ice Normal Operating Range (Green Arc)												14 to 18 amps
Fuel Quantity Indicators No Take-off (Yellow Arc)												0 to 25 gals

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AIRSPEED LIMITATIONS (CAS)

Maximum Allowable (Red Radial) (Gli	de or	Div	re, S	mo	oth	Air)					. 270 mph/235 kts
Caution Range (Yellow Arc)								. 2				mph/208 to 235 kts
Normal Operating Range (Green Arc)									98	to	240	mph/85 to 208 kts
Flap Operating Range (White Arc) .									88	to	156	mph/76 to 135 kts
Approach Position - 15°												. 202 mph/175 kts
Full Down Position - 30°												. 156 mph/135 kts
Single-Engine Best Rate-of-Climb (Blue	Rad	ial)										. 129 mph/112 kts
Minimum Single-Engine Control Speed	(Rec	l Ra	dial)								. 101 mph/88 kts
Maximum Gear Operation Speed .												. 202 mph/175 kts
Maximum Gear Extended Speed .										•		. 202 mph/175 kts
Maximum Design Maneuvering												. 186 mph/161 kts
Maximum Demonstrated Crosswind												. 29 mph/25 kts

ALTITUDE LIMITATION

30,000 ft

MANEUVERS

This is a normal category aircraft. Acrobatic maneuvers, including spins, are prohibited.

Maximum slip duration																			30 seconds
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FLIGHT LOAD FACTORS

At design Gross Weight of 6775 lbs: Positive; Flaps Up 3.5 G, Flaps Down 2.0 G.

CENTER OF GRAVITY (Landing Gear Extended)

Forward Limits: 128.0 inches aft of datum at 5100 lbs and under, then straight line variation to 134.6 inches aft of datum at gross weight of 6775 lbs.

Aft Limits: 139.2 inches aft of datum at all weights.

WEIGHTS

Maximum Take-off Weight				6775 lbs
Maximum Landing Weight (Serials P-190, P-192 and after)				6775 lbs
Maximum Landing Weight (Serials P-4 through P-189 and P-191) Reference Service Instruction No. 0536-202				
8-ply Tires with original unmodified shock strut orifice P/N 60-810012-13 LH, 60-810012-14 RH, or lower dash numbers				6600 lbs
10-ply Tires with original unmodified shock strut orifice P/N 60-810012-13 LH, 60-810012-14 RH, or lower dash numbers				6450 lbs
10-ply Tires with modified or replacement shock strut orifice P/N 60-810012-15 LH, 60-810012-16 RH, or higher dash numbers				6775 lbs
8-Ply Tires with modified or replacement shock strut orifice P/N 60-810012-15 LH, 60-810012-16 RH, or higher dash numbers				6600 lbs
Maximum Ramp Weight	•			6819 lbs

FAA Approved Revised: January 11, 1974

CABIN PRESSURIZATION

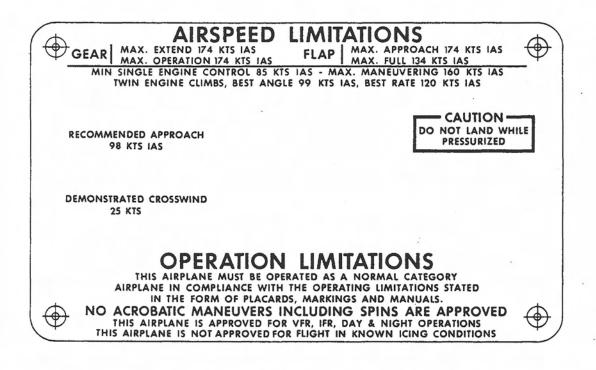
Maximum operating cabin pressure differential is 4.7 psi. Fuselage pressure vessel structural life limit - 15,000 hrs.

AFT FACING CHAIRS

Only aft facing seats are authorized in the aft facing position. The headrest and seat back of the aft facing seat must be in the fully raised position for take-off and landing.

PLACARDS

On the right side wall:



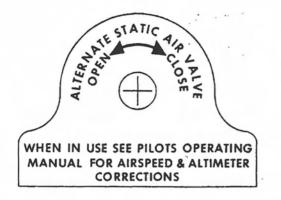
Airplanes equipped in accordance with Airplane Flight Manual Supplement FLIGHT IN KNOWN ICING CONDITIONS, P/N 60-590001-17 are approved for flight in known icing conditions and the following placard will be placed on the Operation Limitation panel:

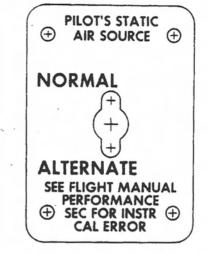


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On these airplanes having Kit 60-5019-1 installed

On the copilot's sidewall: (P-4 through P-246)

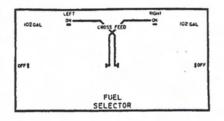




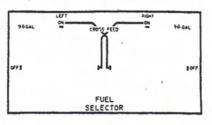
On Fuel Selector Panel on Floor Between Seats:

P-4 through P-195

Unbaffled Fuel Cells

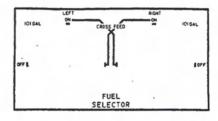


Unbaffled Fuel Cells after compliance with S.I. 0559-281

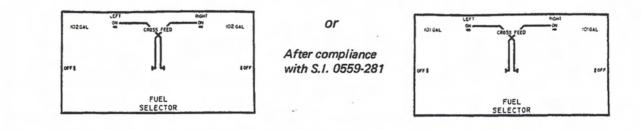


OR

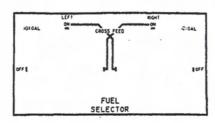
Baffled Fuel Cells both sides per S.I. 0559-281



P-196 through P-219



P-220 and after



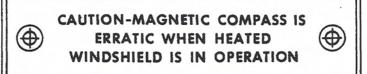
FAA Approved Revised: June 5, 1974

On the main spar cover between the pilot and copilot seats:



Above magnetic compass (Duke 60):

Above magnetic compass (Duke A60):





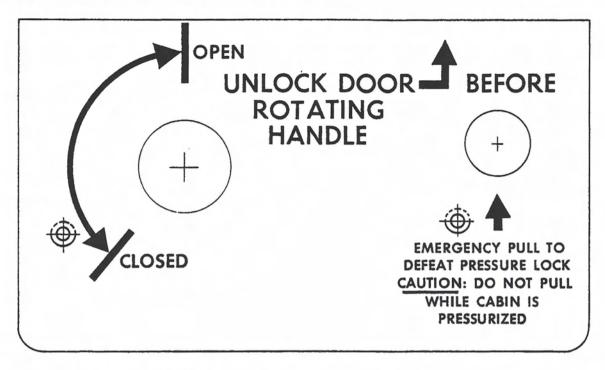
Around each oxygen outlet if installed:



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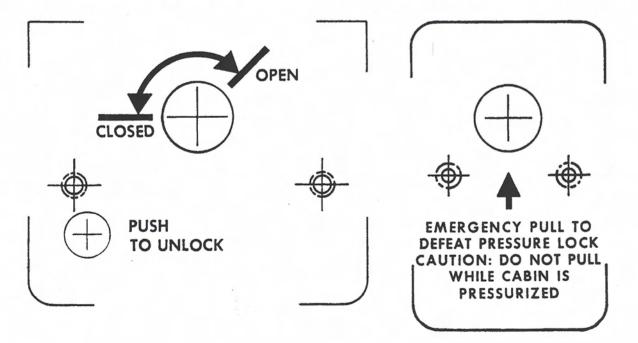
Adjacent to inside cabin door handle (Duke 60):



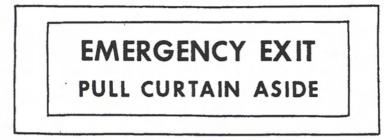
Adjacent to cabin door handle (Duke 60):

Adjacent to inside cabin door handle (Duke A60):

Near cabin door handle (Duke A60):



On headliner above emergency exit window:



On emergency exit:

EMERGENCY EXIT LIFT COVER, RELEASE CATCH

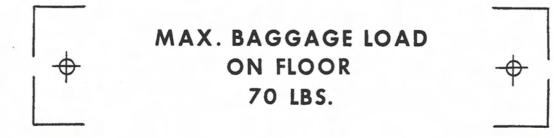
On cabin sidewalls (Duke A60):

HEADRESTS ON AFT FACING SEATS MUST BE EXTENDED UP TO STOPS ON TAKEOFF AND LANDING

In nose baggage compartment:

COMPARTMENT LOADING WEIGHT LIMITATIONS	7
+ STD LOADING 500 LBS MAX.	Φ
SEE WEIGHT & BALANCE SECTION OF FLIGHT MANUAL FOR ADD'L LIMITATIONS FROM OPTIONAL EQUIP.	

On aft cabin floor with optional fifth and sixth seats installed:



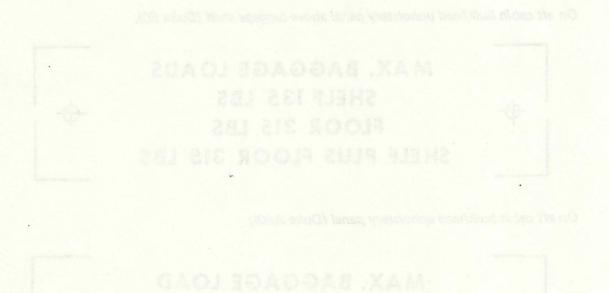
On aft cabin bulkhead upholstery panel above baggage shelf (Duke 60);



On aft cabin bulkhead upholstery panel (Duke A60):



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