

Beechcraft

DUKE B60

OCT. 1978

SPECIFICATIONS



A four to six place, twin engine, high performance, turbocharged, pressurized airplane. Standard equipment includes avionics and a full complement of engine and flight instruments.

Licensed in normal category.

Designed and tested in excess of the required load factors.

Performance

MAXIMUM SPEED246 kts. (283 mph)

CRUISING SPEEDS (Average Cruise Weight)

| | |
|--|--------------------|
| Maximum Cruise Power (Approx. 78% MCP) | |
| 25,000 ft. | 239 kts. (275 mph) |
| 20,000 ft. | 231 kts. (266 mph) |
| 15,000 ft. | 220 kts. (253 mph) |
| Recommended Cruise Power (Approx. 74% MCP) | |
| 25,000 ft. | 233 kts. (268 mph) |
| 20,000 ft. | 225 kts. (259 mph) |
| 15,000 ft. | 214 kts. (246 mph) |
| Recommended Cruise Power (Approx. 68% MCP) | |
| 25,000 ft. | 225 kts. (259 mph) |
| 20,000 ft. | 215 kts. (247 mph) |
| 15,000 ft. | 205 kts. (236 mph) |
| Recommended Cruise Power (Approx. 63% MCP) | |
| 25,000 ft. | 217 kts. (250 mph) |
| 20,000 ft. | 209 kts. (241 mph) |
| 15,000 ft. | 199 kts. (229 mph) |

CRUISE RANGE FOR 232 GALLONS USABLE

(Includes allowance for Fuel Used During Start, Taxi, Take-Off, Climb and a 45 Minute Reserve at Economy Cruise Power; Weight Before Engine Start 6,819 lbs.)

| | |
|--|-----------------------------|
| Maximum Cruise Power (Approx. 78% MCP) | |
| 25,000 ft. | 1,045 naut. mi. (1,203 mi.) |
| 20,000 ft. | 967 naut. mi. (1,113 mi.) |
| 15,000 ft. | 916 naut. mi. (1,054 mi.) |
| Recommended Cruise Power (Approx. 74% MCP) | |
| 25,000 ft. | 1,072 naut. mi. (1,234 mi.) |
| 20,000 ft. | 1,010 naut. mi. (1,163 mi.) |
| 15,000 ft. | 964 naut. mi. (1,110 mi.) |
| Recommended Cruise Power (Approx. 68% MCP) | |
| 25,000 ft. | 1,112 naut. mi. (1,280 mi.) |
| 20,000 ft. | 1,070 naut. mi. (1,232 mi.) |
| 15,000 ft. | 1,028 naut. mi. (1,183 mi.) |
| Recommended Cruise Power (Approx. 63% MCP) | |
| 25,000 ft. | 1,168 naut. mi. (1,344 mi.) |
| 20,000 ft. | 1,122 naut. mi. (1,291 mi.) |
| 15,000 ft. | 1,083 naut. mi. (1,247 mi.) |

RATE OF CLIMB AT SEA LEVEL

| | |
|-------------------------------|-----------|
| Two engines—6,775 lbs. | 1,601 fpm |
| Two engines—6,000 lbs. | 1,930 fpm |
| Two engines—5,200 lbs. | 2,373 fpm |
| Single engine—6,775 lbs. | 307 fpm |
| Single engine—6,000 lbs. | 496 fpm |
| Single engine—5,200 lbs. | 739 fpm |

STALL SPEED — 6,775 lbs. (IAS)

| | |
|-----------------------------|------------------|
| Power off, flaps down | 73 kts. (84 mph) |
| Power off, flaps up | 81 kts. (93 mph) |

SERVICE CEILING

| | |
|--|------------|
| Two engines (100 fpm)—6,775 lbs. | 30,000 ft. |
| Single engine (50 fpm)—6,775 lbs. | 15,100 ft. |
| Single engine (50 fpm)—6,000 lbs. | 20,200 ft. |
| Single engine (50 fpm)—5,200 lbs. | 23,800 ft. |

TAKE-OFF DISTANCE — FAA APPROVED (Normal Procedure at 6,775 lbs.)

| | |
|---|-----------|
| Ground run | 2,075 ft. |
| Total distance over 50 ft. obstacle | 2,626 ft. |

LANDING DISTANCE — FAA APPROVED (Normal Procedure at 6,775 lbs.)

| | |
|---|-----------|
| Ground roll | 1,318 ft. |
| Total distance over 50 ft. obstacle | 3,065 ft. |

The above performance figures are based on the indicated weights and are the results of flight tests conducted by Beech Aircraft Corporation under factory-controlled conditions and will vary with individual aircraft and numerous factors affecting flight performance.

Specifications

WEIGHTS

| | |
|--|------------|
| Ramp Weight | 6,819 lbs. |
| Maximum Weight (Take-off and landing) ... | 6,775 lbs. |
| Empty Weight (Includes Unusable Fuel, Full Oil and Standard Avionics) | 4,383 lbs. |
| Useful Load (Standard Airplane) | 2,436 lbs. |

DIMENSIONS

| | |
|------------------------------------|-----------------------------|
| Wing Span | 39 ft. 3 in. |
| Length | 33 ft. 10 in. |
| Height | 12 ft. 4 in. |
| Cabin Length, inside | 142 in. |
| Cabin Width, inside | 50 in. |
| Cabin Door Opening | 47½ in. high x 26½ in. wide |
| Nose Baggage Door Opening | 23½ in. high x 37½ in. wide |
| Cabin Height, inside | 52 in. |

WING AREA AND LOADINGS

| | |
|---------------------|-------------------|
| Wing Area | 212.9 sq. ft. |
| Wing Loading | 31.8 lbs./sq. ft. |
| Power Loading | 8.9 lbs./hp. |

PRESSURIZATION

| | |
|--|-----------------------|
| (4.6 Differential) | Cabin Altitude |
| Actual Aircraft Altitude—10,000 ft. | Sea Level |
| Actual Aircraft Altitude—21,600 ft. | 8,000 ft. |
| Actual Aircraft Altitude—24,800 ft. | 10,000 ft. |

USABLE FUEL

| | |
|--------------------|-------------|
| Standard | 142 gallons |
| Option No. 1 | 202 gallons |
| Option No. 2 | 232 gallons |

OIL CAPACITY6½ gallons

BAGGAGE

| | |
|--|-------------|
| *Front Baggage Compartment, size | 32 cu. ft. |
| Rear Baggage Area, size (4 seats) | 28¼ cu. ft. |
| *Front Baggage Compartment (Weight Limit 45 lbs./sq. ft.) | 500 lbs. |
| Rear Baggage Limits | 315 lbs. |

*Includes 6.5 cubic feet for Standard Avionics.

Standard Equipment

AVIONICS

Collins VHF-251 Main VHF Transceiver with PWC-150 Power Adapter and B3 Com Antenna
 Collins VIR-351 Omni No. 1 Receiver with IND-351 VOR/ILS Indicator and B38 Antenna
 Collins AMR-350 Audio Panel
 Collins Marker Beacon incorporated in AMR-350, Single Set Marker Lights and B16 Antenna
 Collins ADF-650 ADF with IND-650 Indicator and ANT-650A Antenna
 Collins GLS-350 Glideslope Receiver with A-326A Antenna
 Collins TDR-950 Transponder with B18 Antenna
 Beech Metal Radio Panel, Radio Accessories, and Static Wicks
 Mic Key Button in Pilot's Control Wheel
 White Lighting
 Dual Microphones and Headsets, Single Cockpit Speaker
 Avionics Master Switch

ENGINES AND EQUIPMENT

Two Lycoming 6-cylinder TIO-541-E1C4 Turbo-charged Engines rated at 380 hp each, 2900 rpm for takeoff and continuous operation.
 Propellers — 74" Diameter, Three Blade Aluminum Alloy, Constant Speed, Full Feathering with Hydraulic Governor
 Starters
 Fuel Boost Pumps
 Exhaust Manifolds (stainless steel)
 Pressure Pumps
 Non-Congeaing Oil Radiators
 Full-Flow Oil Filters
 Sonic Choke Venturis for Cabin Pressurization
 Induction Air Filters

LANDING GEAR AND BRAKES

Tricycle type with swiveling steerable nose wheel equipped with shimmy damper. Beech oil-air struts on all wheels designed for smooth taxiing and to withstand the shock created by landing with a vertical descent component of 600 feet per minute. Individual down locks on main gear.
 Main Tires, 19.5 x 6.75-8 10 P.R. (Tube type)
 Nose Wheel Tire, 15 x 6 (Tube type)
 Three Spot Hydraulic Brakes

CONTROLS

Conventional 3-Control System (Pilot's and Co-Pilot's)
 Individual Toe Operated Brakes (Left Side)
 Adjustable Rudder Pedals
 Parking Brakes with Hand Control
 Ventilation, Heating and Pressurization Controls
 Aileron, Rudder and Elevator Trim Tabs Adjustable by Control Wheels
 Power Plant Controls in Console
 Electric Landing Gear and Flap Controls
 Auxiliary Landing Gear Extension Control
 Electric Cowl Flap Controls
 Alternate Static Air Source and Line Drain

ENGINE INSTRUMENTS

Dual Electric Tachometer
 Dual Manifold Pressure Gauge
 Dual Fuel Flow Gauge
 Two Fuel Quantity Gauges
 Two Load Meters
 Dual Turbine Inlet Temperature Gauge
 Two Engine Gauge Groups (with Cylinder Head Temperature, Oil Pressure and Oil Temperature Indicators)

FLIGHT INSTRUMENTS

Airspeed Indicator
 Sensitive Altimeter
 Rate of Climb Indicator
 Gyro Horizon
 Directional Gyro
 Turn Coordinator (electric)
 Compass
 Outside Air Temperature Gauge
 Clock — 8 Day, Control Wheel
 Pressure Gauge
 Flap Position Lights

ELECTRICAL EQUIPMENT

(24-Volt System)

Two 125 Amp. Generators
 Battery — Two 12-Volt (25 Ampere-Hour, 5 hr. rate)
 Two Voltage Regulators
 Electric motors for operating flaps and landing gear
 Electric Actuators for Operating Cowl Flaps
 Heated Pitot Tube
 Heater Blower (Ventilation)
 Heater Blower (Combustion Air)
 Auxiliary Fuel Boost Pumps
 External Power Receptacle
 Fuel Vent Anti-icer (Electric)
 Heated Stall Warning

LIGHTS

Two Landing Lights (One on each Main Gear)
 Position Lights
 Dual Streamline Rotating Beacons
 Steerable Nose Wheel Taxi Light
 Map Light
 Landing Gear Position Lights
 Reading Light at Each Seat
 Cabin Dome Light
 Instrument Post Lights
 Instrument Flood Lighting
 System Annunciator Lights
 Entrance Light
 Front Baggage Compartment Light

CABIN EQUIPMENT

Fresh Air Installation with Provisions to Accommodate Air Conditioning
 Tinted, Cabin Side Windows
 Super Soundproofing
 Full Upholstery including "Wall to Wall" Carpeting
 Two Adjustable Sun Visors
 In-Flight Cabin Storage Pockets
 Ash Tray for Each Seat
 Cigarette Lighter in passenger compartment plus one for pilot and copilot
 45,000 BTU Heater
 Openable Bad-Weather Window for pilot on left
 Coat Hanger Rod
 Arm Rests
 Four reclining, track mounted seats adjust fore and aft for added comfort
 Shoulder Harness on all Forward Facing Seats
 Headrest and Lap Belt provided for all seats
 Baggage Strap Installation for rear of cabin

CABIN PRESSURIZATION EQUIPMENT

Cabin Altitude Selector
 Combined Cabin Altitude and Differential Gauge
 Cabin Rate-of-Climb Indicator
 Cabin Outflow Control Valve
 Cabin Pressurization Safety Valve

Standard Equipment

(Continued)

SERVICE EQUIPMENT

Tow Bar
Service Information Kit
Coat Hanger
Pitot Tube Cover
Sump Drain Wrench
Three Jack Pads
Beechcraft Warranty ID Card
Airplane Log Book
Two Engine Log Books
Control Lock Assembly
Pilot's Check List
Pilot's Operating Manual

SPECIAL FEATURES

Single Point Fueling — Each Wing
Fuel Sight Gage — Each Wing
Quick Removable Gas Tank Caps
Inboard Main Wheel Doors close when wheels are
down to keep out mud and dirt and prevent
buffeting damage
Chair Reclining Mechanism allows vertical to hori-
zontal adjustment on three passenger chairs
Exterior Step Actuates with Landing Gear
Annunciator Panel
Static Wicks
Complete Exterior Polyurethane Paint
Emergency Locator Transmitter



Beech Aircraft Corporation
Wichita, Kansas 67201, U.S.A.

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All specifications and performance data contained in this brochure are subject to change without notice.
Printed in U.S.A. 79SS/B60/A

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